

**Warsaw University  
of Technology**



**Faculty of Power and  
Aeronautical Engineering**

WARSAW UNIVERSITY OF TECHNOLOGY

Institute of Aeronautics and Applied Mechanics

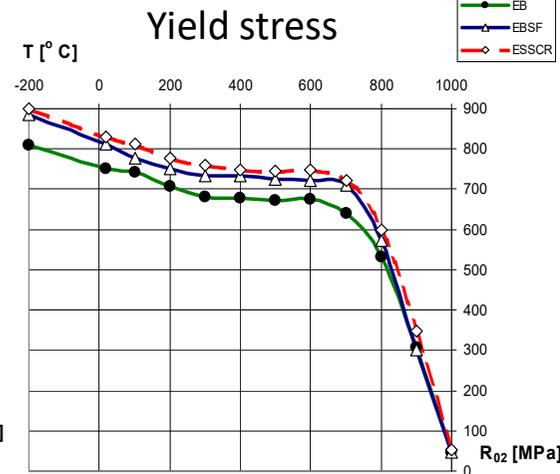
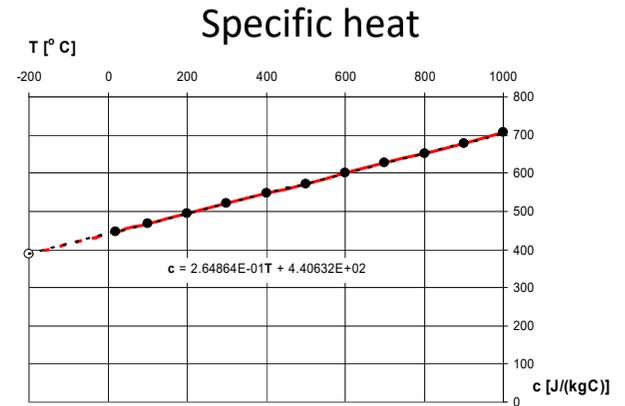
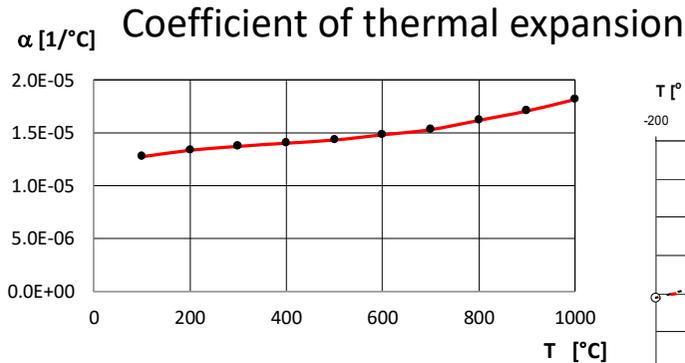
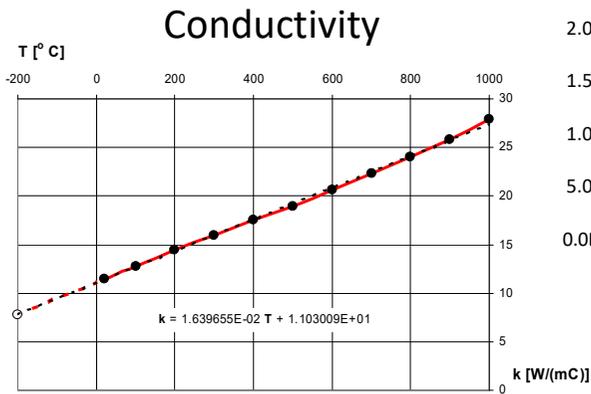
# Finite element method 2 (FEM 2)

Heat transfer and thermal stresses

10.2021

# Thermo-mechanical properties versus temperature

Nimonic 90: nickel-based high-temperature low creep alloy for use in aircraft and gas turbine components such as turbine blades and engine exhaust jets.



EB- extruded bar  
 EBSC - extruded bar subsequently forged  
 ESSCR – extruded section subsequently cold rolled

[www.specialmetals.com](http://www.specialmetals.com)

# Secant coefficient of thermal expansion $\alpha_{se}(T)$

Thermal strain:

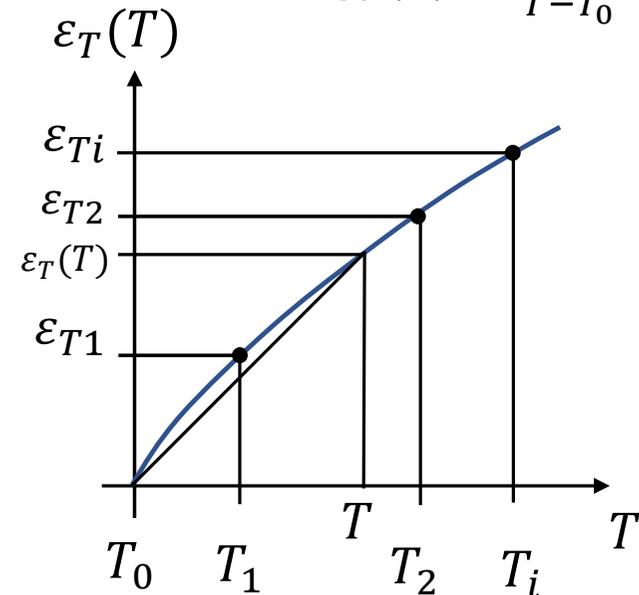
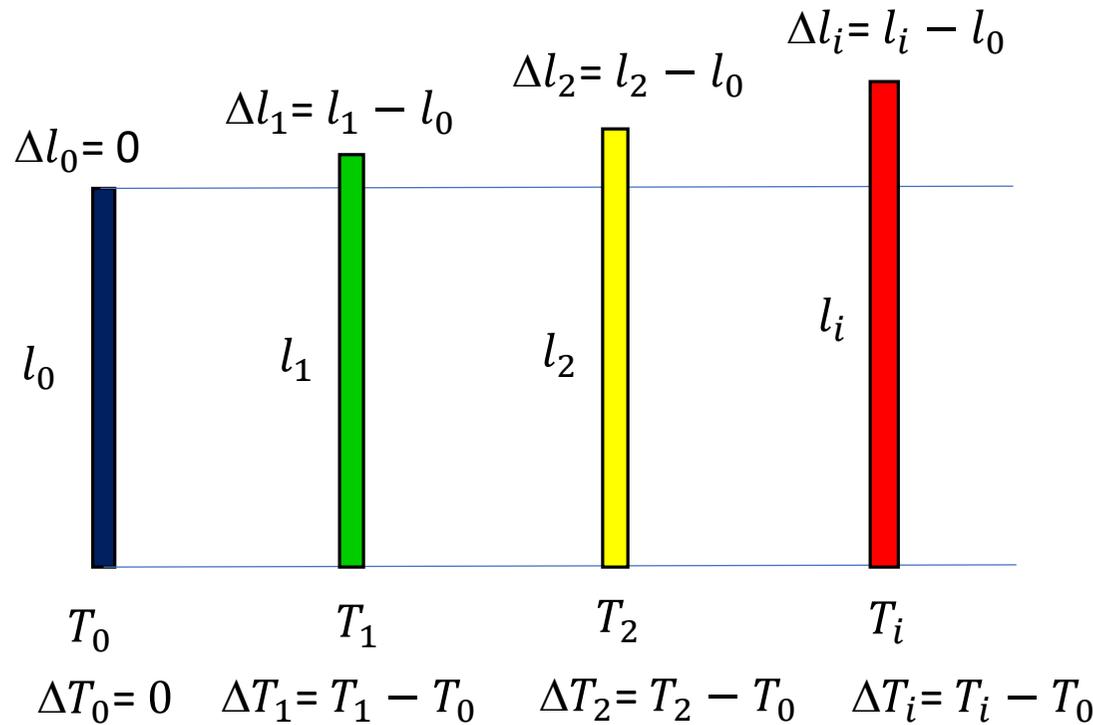
$$\varepsilon_T = \frac{\Delta l}{l} = \alpha_{se} \cdot \Delta T$$

secant coefficient of thermal expansion

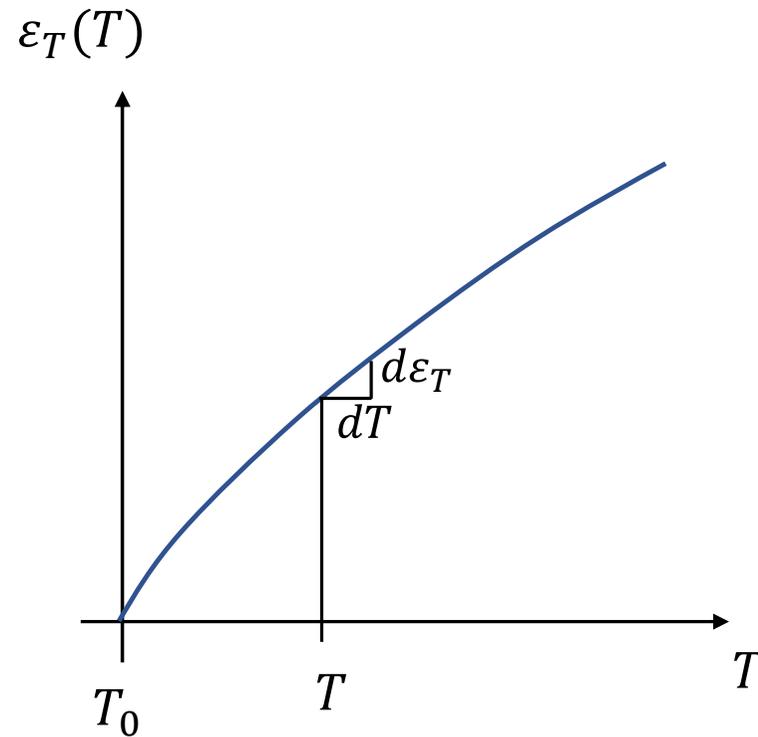
$$\varepsilon_{Ti} = \frac{\Delta l_i}{l_0}$$

$$\alpha_{se}(T_i) = \frac{\varepsilon_{Ti}}{\Delta T_i} = \frac{\Delta l_i}{l_0 \Delta T_i}$$

$$\alpha_{se}(T) = \frac{\varepsilon_T(T)}{T - T_0}$$



Instantaneous coefficient of thermal expansion  $\alpha_{ins}(T)$



$$\alpha_{ins}(T) = \frac{d\varepsilon_T}{dT}$$

## Coefficient of thermal expansion

$T_0$  - temperature at which  $\varepsilon_T = 0$  - in the test

$T_{REF}$  - temperature at which  $\varepsilon_T = 0$  - for working conditions

for  $T_{REF} = T_0$  :

$$\alpha_{se}(T) = \frac{1}{(T - T_0)} \int_{T_0}^T \alpha_{ins}(\bar{T}) d\bar{T}$$

$$\varepsilon_T(T) = \alpha_{se}(T) \cdot (T - T_0)$$

if  $T_{REF} \neq T_0$ , the secant coefficient  $\alpha_{se}(T)$  is recalculated

## Strain in a structure including the thermal effect

The total strain is a sum of thermal and elastic strain:

$$\{\varepsilon\}_{6 \times 1} = \{\varepsilon\}_T_{6 \times 1} + \{\varepsilon\}_e_{6 \times 1} = \begin{Bmatrix} \alpha_x \Delta T \\ \alpha_y \Delta T \\ \alpha_z \Delta T \\ 0 \\ 0 \\ 0 \end{Bmatrix}_T + \begin{Bmatrix} \varepsilon_x \\ \varepsilon_y \\ \varepsilon_z \\ \gamma_{xy} \\ \gamma_{yz} \\ \gamma_{zx} \end{Bmatrix}_e = \begin{Bmatrix} \alpha_x \Delta T \\ \alpha_y \Delta T \\ \alpha_z \Delta T \\ 0 \\ 0 \\ 0 \end{Bmatrix}_T + [D]_{6 \times 6}^{-1} \begin{Bmatrix} \sigma_x \\ \sigma_y \\ \sigma_z \\ \tau_{xy} \\ \tau_{yz} \\ \tau_{zx} \end{Bmatrix}$$

For isotropy:

$$\{\varepsilon\}_{6 \times 1} = \alpha_{se} \cdot \Delta T \begin{Bmatrix} 1 \\ 1 \\ 1 \\ 0 \\ 0 \\ 0 \end{Bmatrix}_T + [D]_{6 \times 6}^{-1} \cdot \{\sigma\}_{6 \times 1}$$

$$\{\varepsilon\}_e_{6 \times 1} = [D]_{6 \times 6}^{-1} \{\sigma\}_{6 \times 1}$$

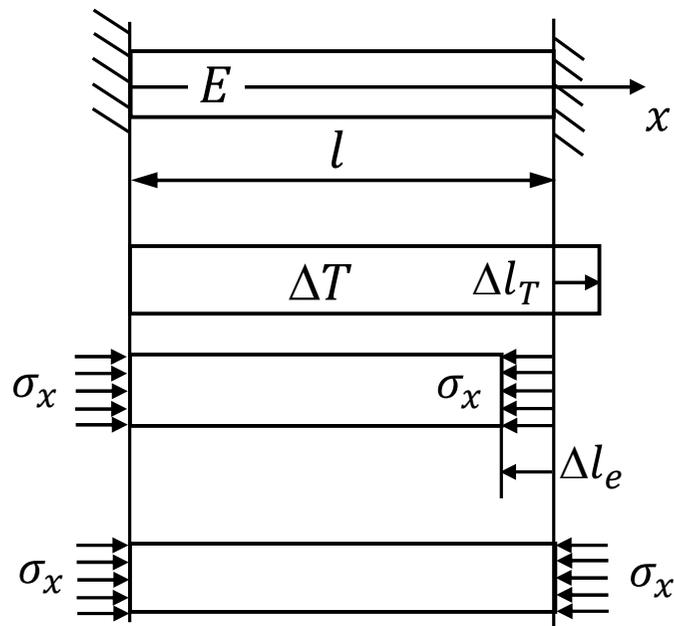
stress vector

inverse constitutive matrix

## Thermal stress is observed for:

- nonuniform temperature field
- temperature change and nonhomogeneous material
- temperature change and statically indeterminate constraints

Example: statically indeterminate bar with uniform temperature distribution



$$\Delta l_T + \Delta l_e = 0$$

$$\varepsilon_{xT} \cdot l + \varepsilon_{xe} \cdot l = 0$$

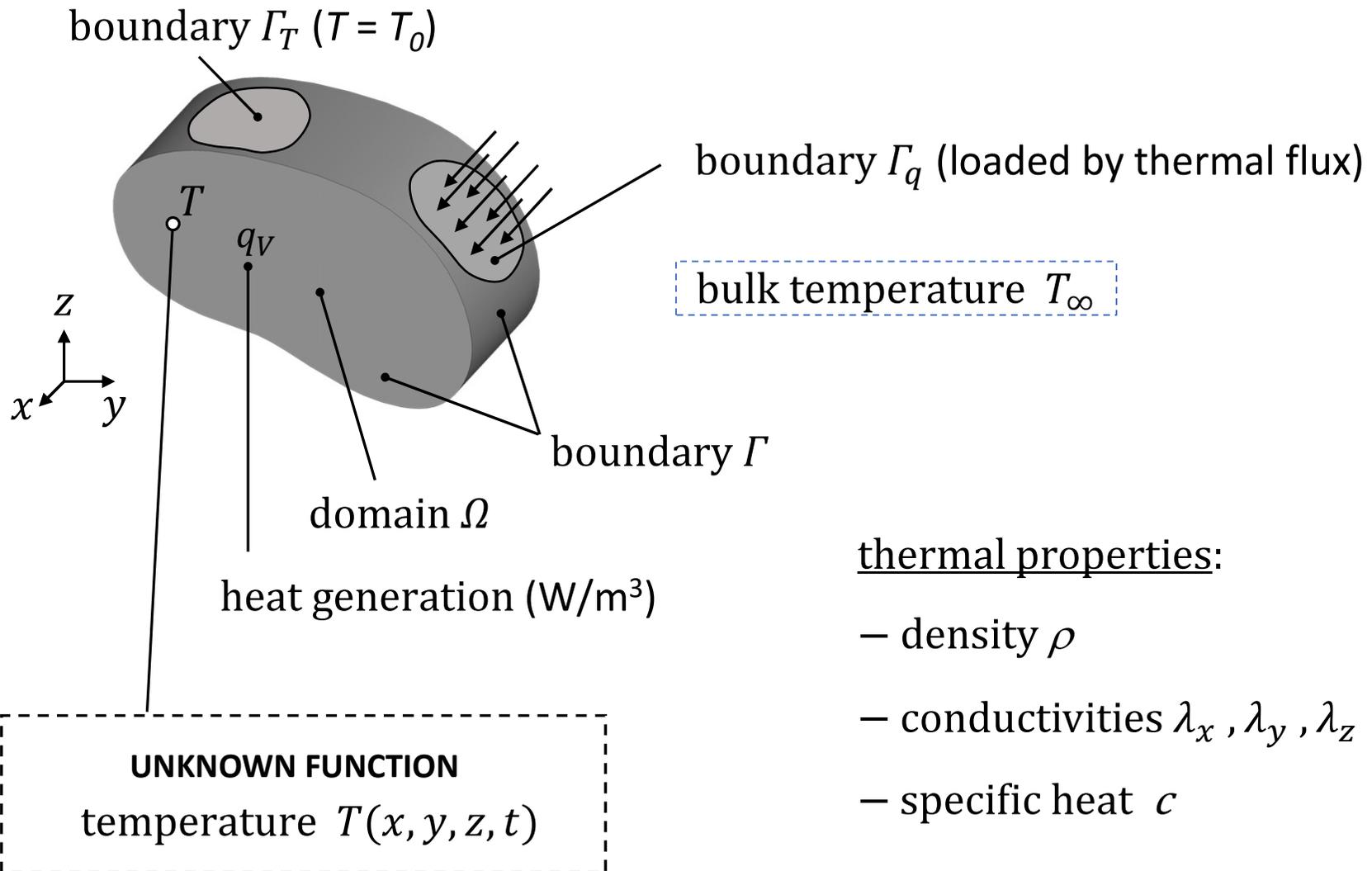
$$\alpha_{se} \cdot \Delta T \cdot l + \frac{\sigma_x}{E} \cdot l = 0$$

$$\sigma_x = -E \cdot \alpha_{se} \cdot \Delta T$$

for:  $E = 2 \cdot 10^5 \text{ MPa}$ ,  $\Delta T = 100^\circ\text{C}$ ,  $\alpha_{se} = 1.2 \cdot 10^{-5} \text{ 1/}^\circ\text{C}$ :

$$\sigma_x = -240 \text{ MPa (compression – possible buckling)}$$

## Boundary value problem of heat transfer



## Thermal analysis

Transient heat flow inside a solid body (law of conservation of energy):

$$\frac{\partial}{\partial x} \left( \lambda_x \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left( \lambda_y \frac{\partial T}{\partial y} \right) + \frac{\partial}{\partial z} \left( \lambda_z \frac{\partial T}{\partial z} \right) + q_V(x, y, z, t) = \rho c \frac{\partial T}{\partial t}$$

$$\rho c \frac{\partial T}{\partial t} = 0 \quad \text{for a steady state:}$$

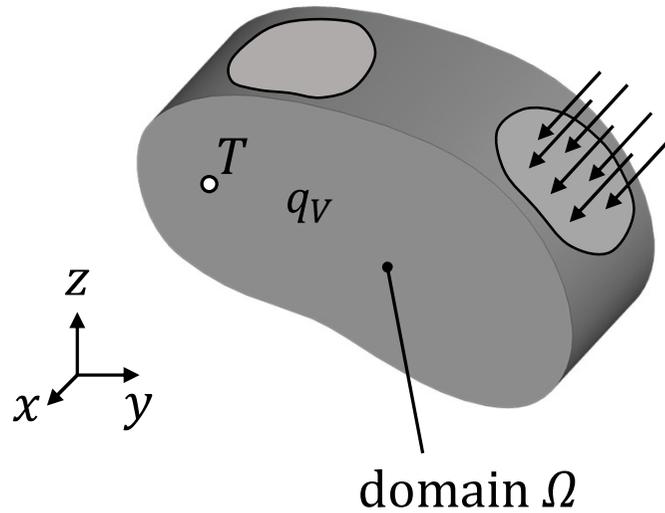
Equivalent functional for steady state analysis :

$$J = \int_{\Omega} \frac{1}{2} \left( \lambda_x \left( \frac{\partial T}{\partial x} \right)^2 + \lambda_y \left( \frac{\partial T}{\partial y} \right)^2 + \lambda_z \left( \frac{\partial T}{\partial z} \right)^2 - 2q_V T \right) d\Omega + \\ + \int_{\Gamma_q} \left( \frac{1}{2} \alpha (T - T_{\infty})^2 + qT \right) d\Gamma_q$$

For isotropy:

$$J = \int_{\Omega} \frac{1}{2} \left( \lambda \left( \left( \frac{\partial T}{\partial x} \right)^2 + \left( \frac{\partial T}{\partial y} \right)^2 + \left( \frac{\partial T}{\partial z} \right)^2 \right) - 2q_V T \right) d\Omega + \int_{\Gamma_q} \left( \frac{1}{2} \alpha (T - T_{\infty})^2 + qT \right) d\Gamma_q$$

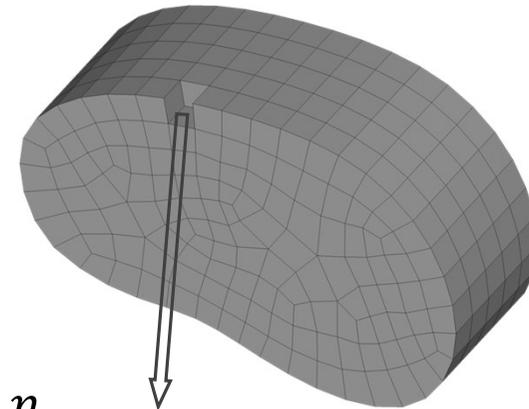
# Finite element model



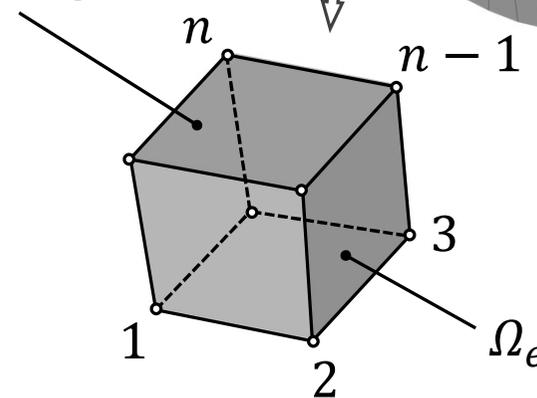
$$\Omega = \sum_{e=1}^{NOE} \Omega_e \text{ and } \Omega_i \cap \Omega_j = 0 \text{ for } i \neq j$$

$NOE$  – no. of finite elements  
 $NON$  – no. of nodes

discretization:



boundary  $\Gamma_{qe}$

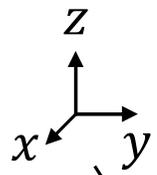
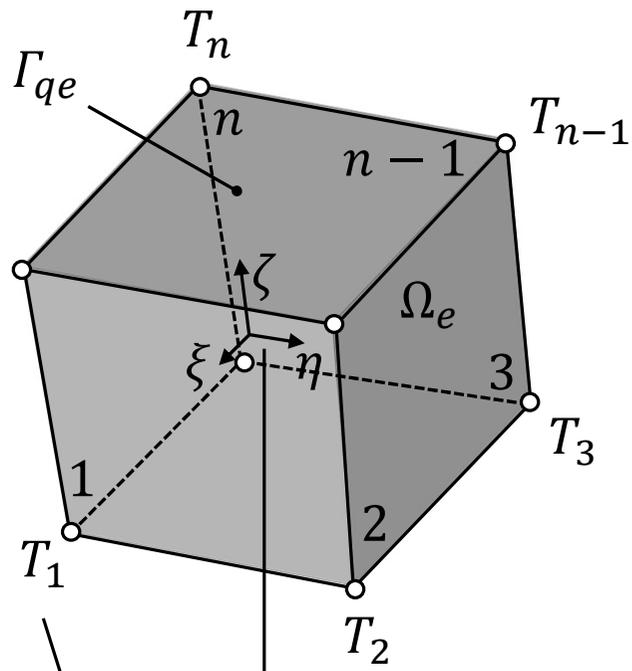


Finite element with  $n$  - nodes

# Nodal approximation in the finite element for thermal analysis

$$\text{temperature } T(\xi, \eta, \zeta) = [N(\xi, \eta, \zeta)] \{T\}_e$$

$\begin{matrix} 1 \times n & n \times 1 \end{matrix}$



coordinate system of a finite element

nodal temperature at node 1

global coordinate system

Vector of shape functions

$$[N(\xi, \eta, \zeta)] = [N_1, N_2, \dots, N_i, \dots, N_n]$$

$1 \times n$

Local vector of nodal temperatures

$$\{T\}_e = \begin{Bmatrix} T_1 \\ T_2 \\ T_3 \\ \vdots \\ T_i \\ \vdots \\ T_n \end{Bmatrix}_e$$

$n \times 1$

Functional formulation for a finite element (isotropy and steady state):

$$J_e = \int_{\Omega_e} \frac{1}{2} \left( \lambda \left( \left( \frac{\partial T}{\partial x} \right)^2 + \left( \frac{\partial T}{\partial y} \right)^2 + \left( \frac{\partial T}{\partial z} \right)^2 \right) - 2q_V T \right) d\Omega_e + \int_{\Gamma_{qe}} \left( \frac{1}{2} \alpha (T - T_\infty)^2 + qT \right) d\Gamma_{qe}$$

derivative of  $J_e$  with respect to temperature  $T_i$ :

$$\frac{\partial J_e}{\partial T_i} = \int_{\Omega_e} \left( \lambda \left( \frac{\partial T}{\partial x} \cdot \frac{\partial}{\partial T_i} \left( \frac{\partial T}{\partial x} \right) + \frac{\partial T}{\partial y} \cdot \frac{\partial}{\partial T_i} \left( \frac{\partial T}{\partial y} \right) + \frac{\partial T}{\partial z} \cdot \frac{\partial}{\partial T_i} \left( \frac{\partial T}{\partial z} \right) \right) - q_V \frac{\partial T}{\partial T_i} \right) d\Omega_e + \int_{\Gamma_{qe}} \left( \alpha (T - T_\infty) \frac{\partial T}{\partial T_i} + q \frac{\partial T}{\partial T_i} \right) d\Gamma_{qe}$$

$$T(\xi, \eta, \zeta) = \underset{1 \times n}{[N(\xi, \eta, \zeta)]} \underset{n \times 1}{\{T\}_e} = [N_1, N_2, \dots, N_i, \dots, N_n] \left\{ \begin{array}{c} T_1 \\ T_2 \\ T_3 \\ \vdots \\ T_i \\ \vdots \\ T_n \end{array} \right\}_e$$

$$\frac{\partial T}{\partial T_i} = N_i$$

$$\frac{\partial T}{\partial x} = \underbrace{\frac{\partial [N]}{\partial x}}_{1 \times n} \underbrace{\{T\}_e}_{n \times 1} + [N] \underbrace{\frac{\partial \{T\}_e}{\partial x}}_{n \times 1} = \left[ \frac{\partial N_1}{\partial x}, \frac{\partial N_2}{\partial x}, \dots, \frac{\partial N_i}{\partial x}, \dots, \frac{\partial N_n}{\partial x} \right] \underbrace{\begin{Bmatrix} T_1 \\ T_2 \\ T_3 \\ \vdots \\ T_i \\ \vdots \\ T_n \end{Bmatrix}}_e$$

$$\uparrow$$

$$\frac{\partial \{T\}_e}{\partial x} = \{0\}$$

$$\frac{\partial N_i}{\partial x} = \frac{\partial N_i}{\partial \xi} \cdot \frac{\partial \xi}{\partial x} + \frac{\partial N_i}{\partial \eta} \cdot \frac{\partial \eta}{\partial x} + \frac{\partial N_i}{\partial \zeta} \cdot \frac{\partial \zeta}{\partial x} \quad \text{(Jacobian matrix)}$$

$$\frac{\partial}{\partial T_i} \left( \frac{\partial T}{\partial x} \right) = \frac{\partial N_i}{\partial x}$$

similarly:

$$\frac{\partial T}{\partial y} = \left[ \frac{\partial N_1}{\partial y}, \frac{\partial N_2}{\partial y}, \dots, \frac{\partial N_i}{\partial y}, \dots, \frac{\partial N_n}{\partial y} \right] \begin{Bmatrix} T_1 \\ T_2 \\ T_3 \\ \vdots \\ T_i \\ \vdots \\ T_n \end{Bmatrix}_e$$

$$\frac{\partial T}{\partial z} = \left[ \frac{\partial N_1}{\partial z}, \frac{\partial N_2}{\partial z}, \dots, \frac{\partial N_i}{\partial z}, \dots, \frac{\partial N_n}{\partial z} \right] \begin{Bmatrix} T_1 \\ T_2 \\ T_3 \\ \vdots \\ T_i \\ \vdots \\ T_n \end{Bmatrix}_e$$

$$\frac{\partial N_i}{\partial y} = \frac{\partial N_i}{\partial \xi} \cdot \frac{\partial \xi}{\partial y} + \frac{\partial N_i}{\partial \eta} \cdot \frac{\partial \eta}{\partial y} + \frac{\partial N_i}{\partial \zeta} \cdot \frac{\partial \zeta}{\partial y}$$

$$\frac{\partial N_i}{\partial z} = \frac{\partial N_i}{\partial \xi} \cdot \frac{\partial \xi}{\partial z} + \frac{\partial N_i}{\partial \eta} \cdot \frac{\partial \eta}{\partial z} + \frac{\partial N_i}{\partial \zeta} \cdot \frac{\partial \zeta}{\partial z}$$

$$\frac{\partial}{\partial T_i} \left( \frac{\partial T}{\partial y} \right) = \frac{\partial N_i}{\partial y}$$

$$\frac{\partial}{\partial T_i} \left( \frac{\partial T}{\partial z} \right) = \frac{\partial N_i}{\partial z}$$

derivative of  $J_e$  with respect to temperature  $T_i$ :

$$\frac{\partial J_e}{\partial T_i} = \int_{\Omega_e} \lambda \left( \frac{\partial [N]}{\partial x} \cdot \frac{\partial N_i}{\partial x} + \frac{\partial [N]}{\partial y} \cdot \frac{\partial N_i}{\partial y} + \frac{\partial [N]}{\partial z} \cdot \frac{\partial N_i}{\partial z} \right) d\Omega_e - \int_{\Omega_e} q_V N_i d\Omega_e$$

$$+ \int_{\Gamma_{qe}} (\alpha([N]\{T\}_e - T_\infty)N_i + qN_i) d\Gamma_{qe}$$

derivatives of  $J_e$  with respect to temperatures  $T_1, T_2, \dots, T_i, \dots, T_n$ :

$$\frac{\partial J_e}{\partial T_1}, \frac{\partial J_e}{\partial T_2}, \dots, \frac{\partial J_e}{\partial T_i}, \dots, \frac{\partial J_e}{\partial T_n}$$



$$\frac{\partial J_e}{\partial \{T\}_e} = [h]_e \{T\}_e + \{F\}_e$$

local conductivity matrix

thermal load vector in a finite element

components of the local conductivity matrix:

$$h_{ij} = \int_{\Omega_e} \lambda \left( \frac{\partial N_i}{\partial x} \cdot \frac{\partial N_j}{\partial x} + \frac{\partial N_i}{\partial y} \cdot \frac{\partial N_j}{\partial y} + \frac{\partial N_i}{\partial z} \cdot \frac{\partial N_j}{\partial z} \right) d\Omega_e$$

components of the thermal load vector:

$$F_{ie} = - \int_{\Omega_e} q_V N_i d\Omega_e + \int_{\Gamma_{qe}} \alpha ([N] \{T\}_e - T_\infty) N_i d\Gamma_{qe} + \int_{\Gamma_{qe}} q N_i d\Gamma_{qe}$$

heat generation
convection & radiation
applied thermal flux

### 3 ways of heat transfer

- conduction

$$q = -\lambda \cdot \text{grad}(T)$$

rate of heat flow per unit area (HEAT FLUX)

- convection

$$q = \alpha(T - T_{\infty})$$

film coefficient

bulk emperature  
surface temperature

- radiation

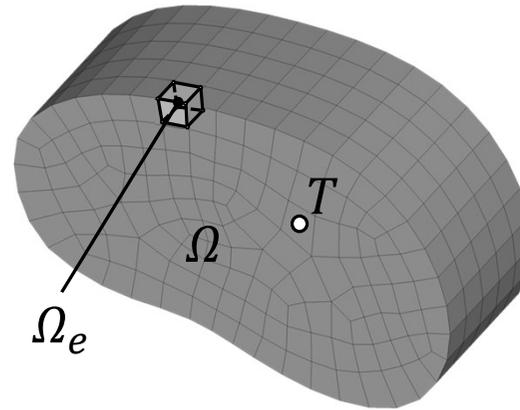
$$\alpha = f(T) = c \frac{T^4 - T_{\infty}^4}{T - T_{\infty}}$$

Medium (fluid)	Free convection	Forced convection
gas (air)	5–30	30–500
water	30–300	300–20000
oil	5–100	30–3000
liquid metals	50–500	500–20000

$$\left( \frac{W}{m^2 K} \right)$$

Minimalization of equivalent functional  $J$  for the entire FE model:

$NOE$  – no. of FEs  
 $NON$  – no. of nodes  
 $NON = NDOF$



To find the solution of heat transfer, the derivative of functional  $J$  with respect to the global vector of nodal temperatures is minimized:

$$\frac{\partial J}{\partial \{T\}} = [H] \cdot \{T\} + \{F\} = \{0\}$$

$NON \times 1$        $NON \times NON$      $NON \times 1$      $NON \times 1$      $NON \times 1$

global conductivity matrix

global vector of nodal temperatures

global thermal load vector

Including boundary conditions:

$$N = NON - NOF$$

 number of known nodal temperatures

Set of algebraic equation after taking boundary conditions into account:

$$\underset{N \times N}{[H]} \cdot \underset{N \times 1}{\{T\}} + \underset{N \times 1}{\{F\}} = \underset{N \times 1}{\{0\}}$$